

Although we stipulated simple, homogeneous shear flow be produced throughout the flow domain, can we, perhaps, relax that requirement?  $\underline{v} \equiv \begin{pmatrix} \dot{\varsigma}(t)x_2 \\ 0 \\ 0 \end{pmatrix}_{123}$   $\underline{v} = \begin{pmatrix} \dot{\varsigma}(t)x_2 \\ 0 \\ 0 \end{pmatrix}_{123}$ 

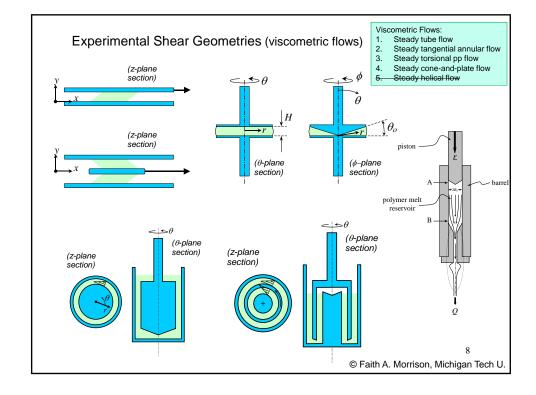
Viscometric flow: motions that are locally equivalent to steady simple shearing motion at every particle

- •globally steady with respect to some frame of reference
- •streamlines that are straight, circular, or helical
- •each flow can be visualized as the relative motion of a sheaf of material surfaces (slip surfaces)
- •each slip surface moves without changing shape during the motion
- •every particle lies on a material surface that moves without stretching (inextensible slip surfaces)

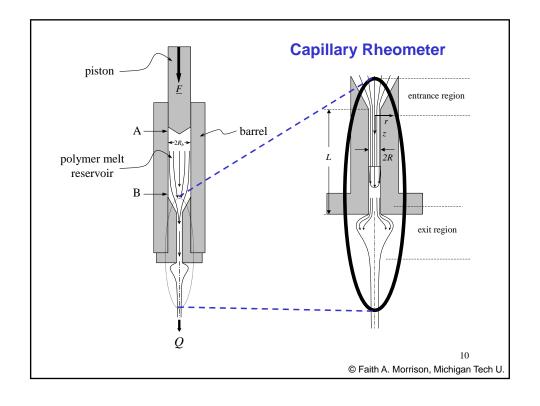
#### Viscometric Flows:

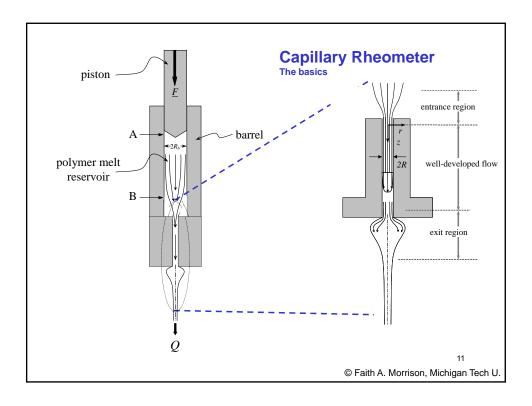
- 1. Steady tube flow
- 2. Steady tangential annular flow
- 3. Steady torsional flow (parallel plate flow)
- 4. Steady cone-and-plate flow (small cone angle)
- 5. Steady helical flow

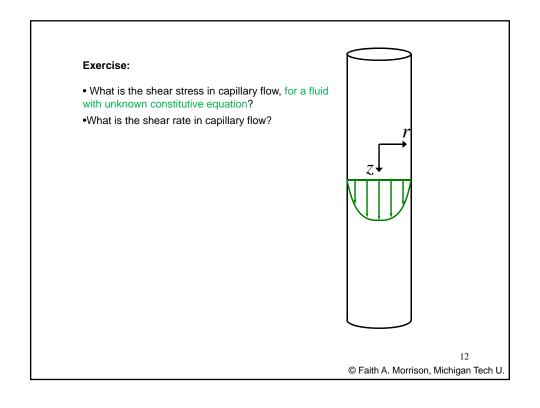
Wan-Lee Yin, Allen C. Pipkin, "Kinematics of viscometric flow," *Archive for Rational Mechanics and Analysis*, 37(2) 111-135, 1970
R. B. Bird, R. Armstrong, O. Hassager, *Dynamics of Polymeric Liquids*, 2<sup>nd</sup> edition, Wiley



# Types of Shear Rheometry **Mechanical:** •Mechanically produce linear drag flow; Measure (shear strain transducer): 1. planar Couette Shear stress on a surface •Mechanically produce torsional drag flow; 1. cone and plate; Measure: (strain-gauge; force rebalance) parallel plate; circular Couette Torque to rotate surfaces Back out material functions •Produce pressure-driven flow through conduit capillary flow slit flow Measure: Pressure drop/flow rate Back out material functions © Faith A. Morrison, Michigan Tech U.







To calculate shear rate, shear stress, look at EOM:

$$\eta = \frac{\tau_R}{\dot{\gamma}_R} \qquad P \equiv p - \rho gz$$

$$\rho \left( \frac{\partial \underline{v}}{\partial t} + \underline{v} \cdot \nabla \underline{v} \right) = -\nabla P - \nabla \cdot \underline{\tau} = 0$$

steady unidirectional state

$$\begin{pmatrix} 0 \\ 0 \\ 0 \\ 0 \end{pmatrix}_{r\theta z} = \begin{pmatrix} -\frac{\partial P}{\partial r} \\ 0 \\ -\frac{\partial P}{\partial z} \end{pmatrix}_{r\theta z} - \begin{pmatrix} \frac{1}{r} \frac{\partial r \tau_{rr}}{\partial r} - \frac{\tau_{\theta \theta}}{r} \\ -\frac{1}{r^2} \frac{\partial r^2 \tau_{r\theta}}{\partial r} \\ \frac{1}{r} \frac{\partial r \tau_{rz}}{\partial r} \\ -\frac{\partial P}{\partial z} \end{pmatrix}_{r\theta z} \begin{pmatrix} \frac{1}{r} \frac{\partial r \tau_{rz}}{\partial r} \\ -\frac{1}{r} \frac{\partial r \tau_{rz}}{\partial r} \end{pmatrix}_{r\theta z}$$
 •long tube •symmetric stress tensor •lsothermal •Viscosity indep of pressure

Assume:

- •Incompressible fluid
- •no θ-dependence
- •long tube

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Shear stress in capillary flow:

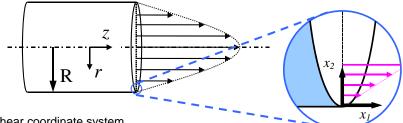
$$\tau_{rz} = \frac{(P_0 - P_L)r}{2L} = \tau_R \frac{r}{R}$$

$$\left(\frac{\partial P}{\partial z} = \text{constant}\right)$$

(varies with position, i.e. inhomogeneous flow)

What was the shear stress in drag flow?

# Viscosity from capillary flow - inhomogeneous shear flow



Shear coordinate system near wall:

$$\hat{e}_1 = \hat{e}_z$$

$$\hat{e}_2 = -\hat{e}_z$$

$$\begin{aligned} \tau_{21} &= -\tau_{rz}|_{r=R} \equiv -\tau_{R} \\ \dot{\gamma}_{0} &= -\frac{\partial v_{z}}{\partial z} = -\frac{\partial v_{z}}{\partial z} = \dot{\gamma}|_{z=0} \equiv \dot{\gamma}_{R} \end{aligned}$$

$$\begin{aligned} \hat{e}_1 &= \hat{e}_z \\ \hat{e}_2 &= -\hat{e}_r \\ \hat{e}_3 &= -\hat{e}_\theta \end{aligned} \qquad \begin{aligned} \tau_{21} &= -\tau_{rz}\big|_{r=R} \equiv -\tau_R \\ \dot{\gamma}_0 &= \frac{\partial v_z}{\partial (-r)} = -\frac{\partial v_z}{\partial r} = \dot{\gamma}\big|_{r=R} \equiv \dot{\gamma}_R \\ \eta &= \frac{-\tau_{21}}{\dot{\gamma}_0} = \frac{\tau_R}{\dot{\gamma}_R} \end{aligned} \qquad \text{wall shear stress wall shear rate}$$

It is not the same shear rate everywhere, but if we focus on the wall we can still get  $\eta(\gamma_R)$ 

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Viscosity from Wall Stress/Shear rate



Note: we are assuming no-slip at the wall

Wall shear stress in capillary flow:

is in capillary flow: 
$$\left. \tau_{rz} \right|_{r=R} = \frac{\left(P_0 - P_L\right)r}{2L} \right|_{r=R} = \frac{\Delta PR}{2L}$$
 
$$\left(\frac{\partial P}{\partial z} = \text{constant}\right)$$

$$\left(\frac{\partial P}{\partial z} = \text{constant}\right)$$

What is shear rate at the wall in capillary flow?

$$\dot{\gamma} = \frac{\partial v_z}{\partial (-r)} = -\frac{\partial v_z}{\partial r} = \dot{\gamma}\Big|_{r=R} \equiv \dot{\gamma}_R$$

If  $v_z(r)$  is known, this is easy to calculate.

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# Velocity fields, Flow in a Capillary

Newtonian fluid: 
$$v_z(r) = \frac{2Q}{\pi R^2} \left[ 1 - \left(\frac{r}{R}\right)^2 \right]$$

# Power-law GNF fluid:

$$v_{z}(r) = R^{\frac{1}{n}+1} \left( \frac{P_{0} - P_{L}}{2mL} \right)^{\frac{1}{n}} \left( \frac{1}{1/n+1} \right) \left[ 1 - \left( \frac{r}{R} \right)^{\frac{1}{n}+1} \right]$$

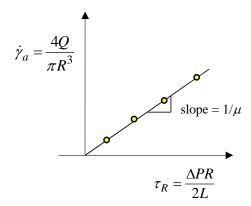
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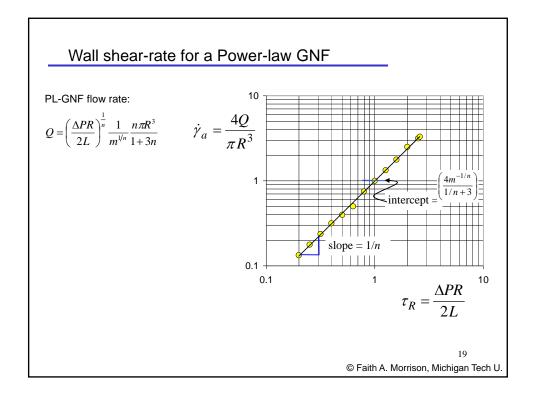
# Wall shear-rate for a Newtonian fluid

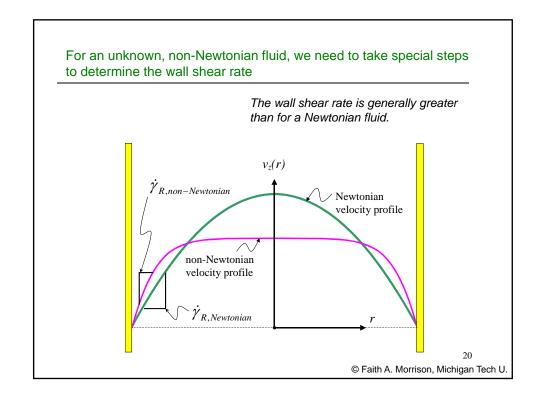
Hagen-Poiseuille:

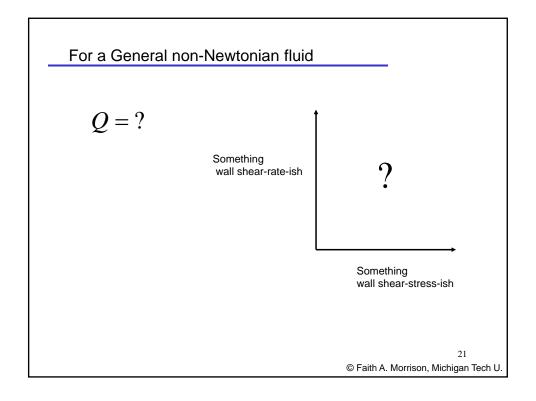
$$Q = \frac{\pi \Delta P R^4}{8\mu L}$$
$$\frac{4Q}{\pi R^3} = \frac{1}{\mu} \frac{\Delta P R}{2L}$$

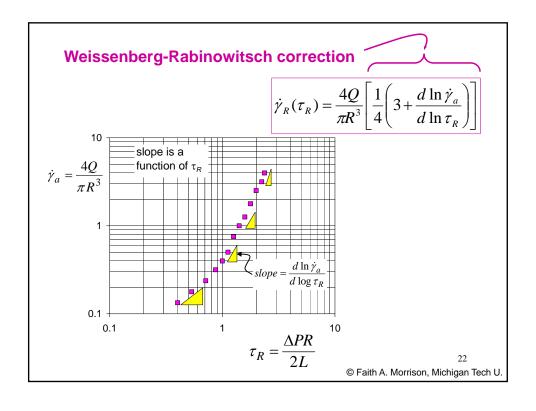


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# Capillary flow

#### Assumptions:

- •Steady.....•No intermittent flow allowed
- $\bullet \theta \ \text{symmetry}.....\bullet \text{No spiraling flow allowed}$
- •Unidirectional .....•Check end effects
- •Incompressible......•Avoid high absolute pressures
- •Constant pressure gradient...•Check end effects
- •No slip.....•Check wall slip

#### Methods have been devised to account for

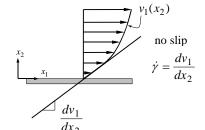
- •Slip
- •End effects

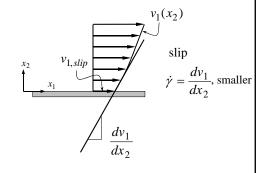
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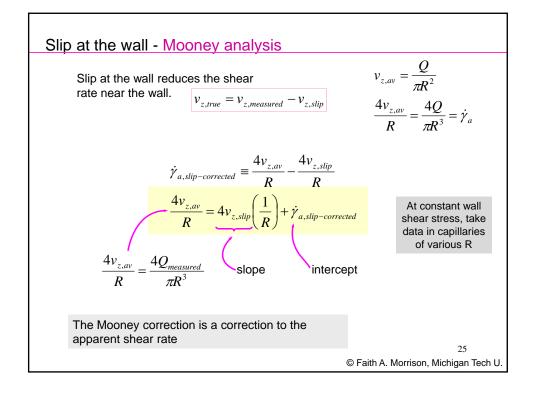
# Slip at the wall - Mooney analysis

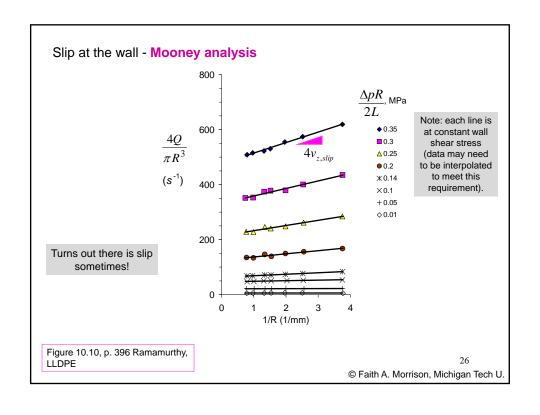
Slip at the wall reduces the shear rate near the wall.

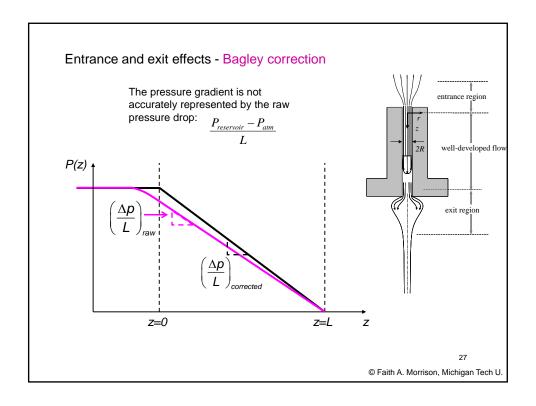


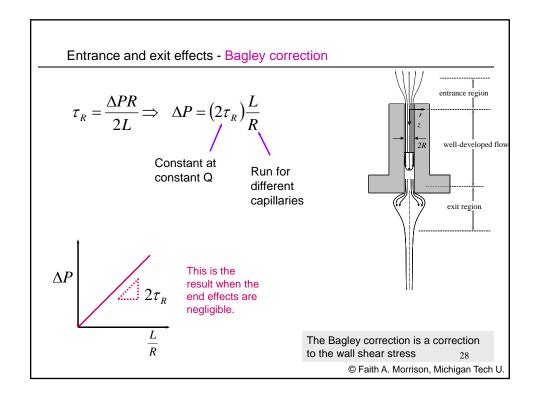


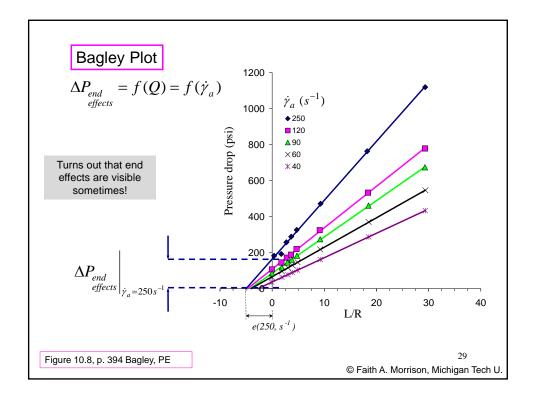
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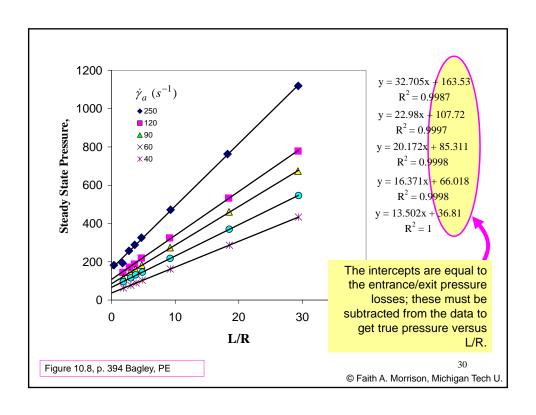


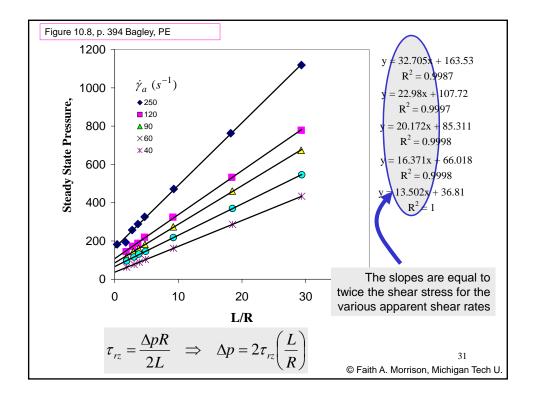


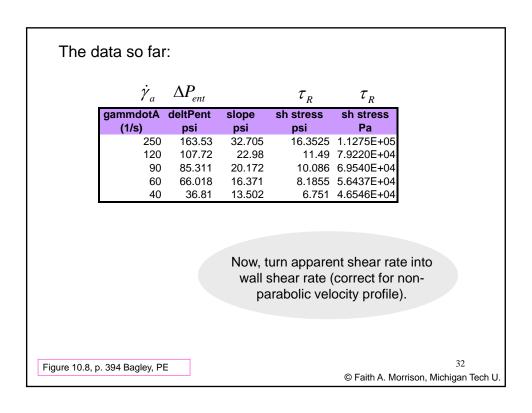


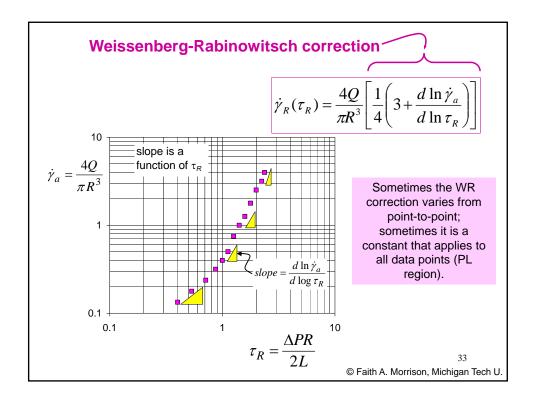


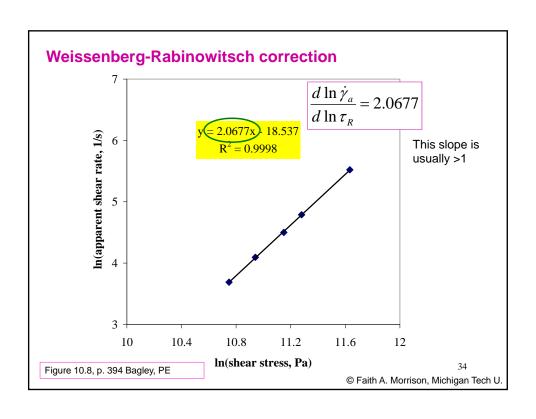


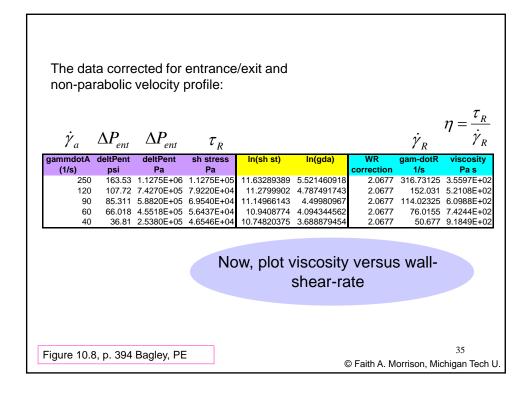


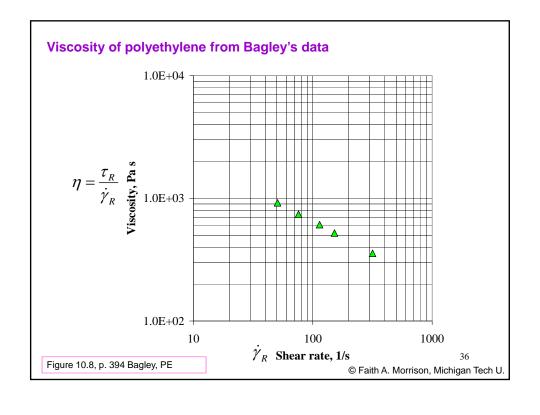










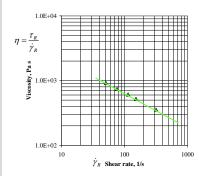


### **Viscosity from Capillary Experiments, Summary:**

- Take data of pressure-drop versus flow rate for capillaries of various lengths; perform Bagley correction on Δp (entrance pressure losses)
- 2. If possible, also take data for capillaries of different radii; perform Mooney correction on Q (slip)
- 3. Perform the Weissenberg-Rabinowitsch correction (obtain correct wall shear rate)
- 4. Plot true viscosity versus true wall shear rate
- 5. Calculate power-law *m*, *n* from fit to final data (if appropriate)

raw data:  $\Delta P(Q)$ 

final data:  $\, \eta = au_{\scriptscriptstyle R} \, / \, \dot{\gamma}_{\scriptscriptstyle R} \,$ 



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What about the shear normal stresses,  $\Psi_1$ ,  $\Psi_2$  from capillary data?

Extrudate swell -relation to  $N_1$  is model dependent (see discussion in Macosko, p254)

$$N_1^2 = 8\tau_R^2 \left( \left( \frac{D_e}{2R} \right)^6 - 1 \right)$$

 $D_e =$  Extrudate diameter

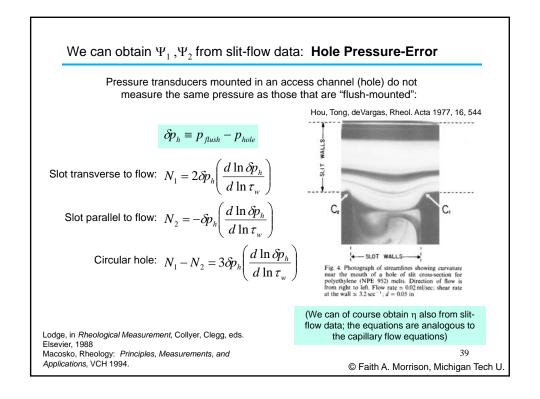
Assuming unconstrained recovery after steady shear, K-BKZ model with one relaxation time

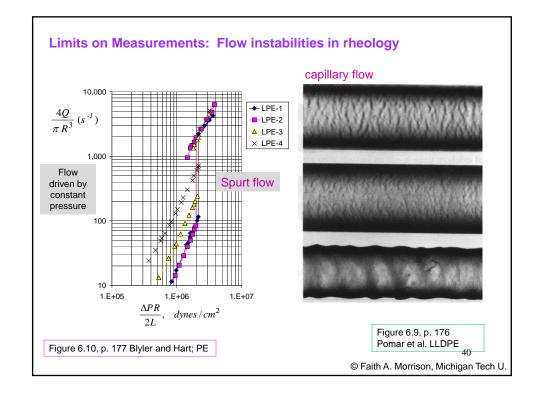
Not a great method; can perhaps be used to index materials

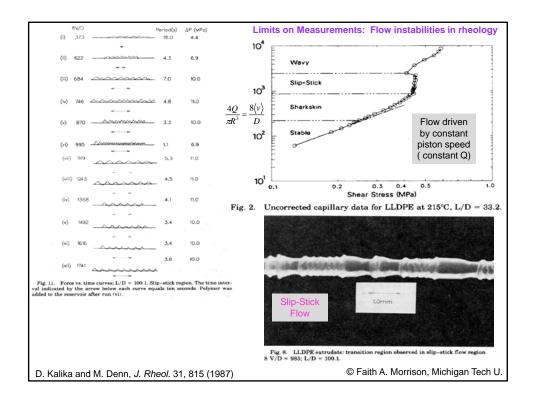
 $\Psi_2$  ? (cannot obtain from capillary flow, but...)

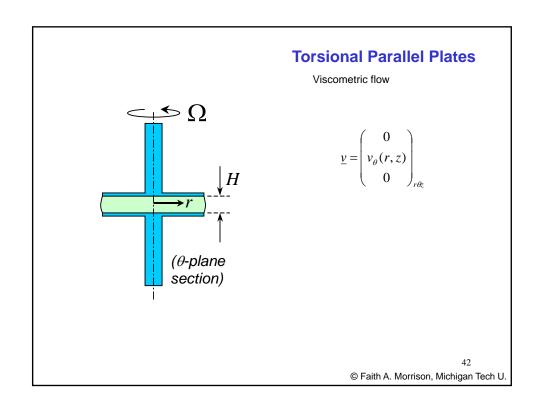
Macosko, Rheology: *Principles, Measurements, and Applications*, VCH 1994.

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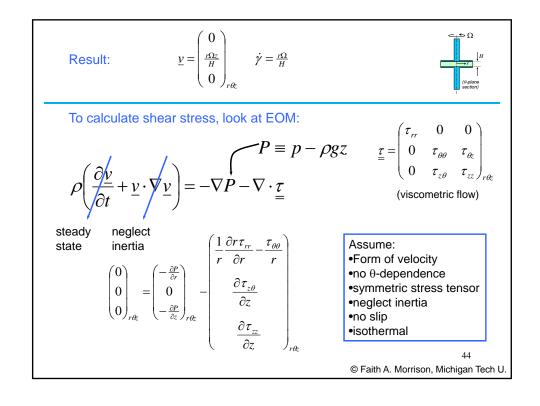


To calculate shear rate: 
$$v_{\theta} = A(r)z + B(r)$$

$$v_{\theta} = \frac{r\Omega z}{H} \qquad \text{(due to boundary conditions)}$$

$$\dot{\gamma} = \left| \dot{\underline{\gamma}} \right| = \begin{bmatrix} 0 & \frac{\partial v_{\theta}}{\partial r} - \frac{v_{\theta}}{r} & 0 \\ \frac{\partial v_{\theta}}{\partial r} - \frac{v_{\theta}}{r} & 0 & \frac{\partial v_{\theta}}{\partial z} \\ 0 & \frac{\partial v_{\theta}}{\partial z} & 0 \end{bmatrix}_{r\theta z}$$

$$\dot{\gamma} = ?$$



#### Result:

$$\frac{\partial \tau_{z\theta}}{\partial z} = 0$$

$$\tau_{z\theta} = f(r)$$

The experimentally measurable variable is the torque to turn the plate:

$$\underline{T} = \iint_{S} \left[ \underline{R} \times \left( \hat{n} \cdot - \underline{\underline{\Pi}} \right) \right]_{surface} dS$$

$$\underline{T} = \iint_{S} \left[ \underline{R} \times \left( \hat{n} \cdot - \underline{\underline{\Pi}} \right) \right]_{surface} dS$$

$$\underline{T} = \iint_{0}^{2\pi R} \left[ r \hat{e}_{r} \times \left( \hat{e}_{z} \cdot - \underline{\underline{\Pi}} \right) \right]_{z=H} r dr d\theta$$

$$T_z = 2\pi \int_{0}^{R} \left[ -\tau_{z\theta} \right]_{z=H} r^2 dr$$

Following Rabinowitsch, replace stress with viscosity, r with shear rate, and differentiate.

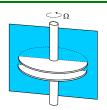
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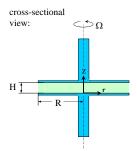
#### Torsional Parallel-Plate Flow - Viscosity

Measureables:

Torque T to turn plate Rate of angular rotation  $\Omega$ 

Note: shear rate experienced by fluid elements depends on their r position. (consider effect on complex fluids)



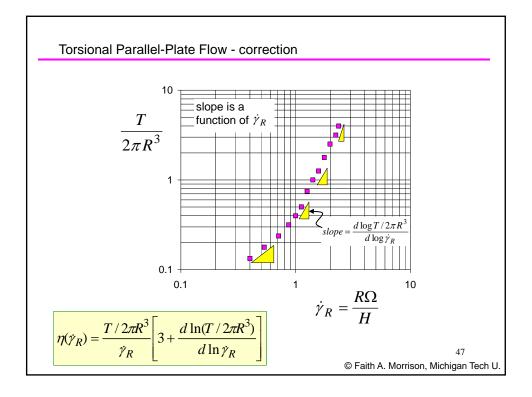


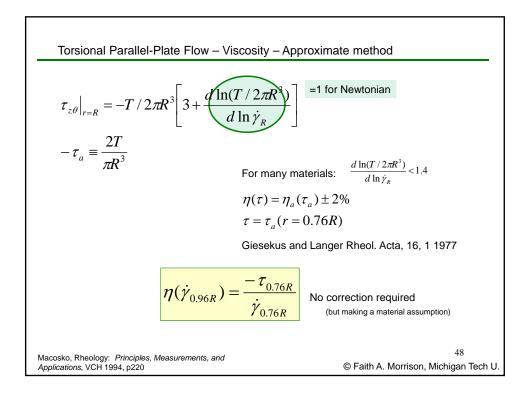
$$\dot{\gamma} = \frac{r\Omega}{H} = \dot{\gamma}_R \frac{r}{R}$$

By carrying out a Rabinowitsch-like calculation, we can obtain the stress at the rim (r=R). 
$$\tau_{z\theta}\big|_{r=R} = -T/2\pi R^3 \left[ 3 + \frac{d\ln(T/2\pi R^3)}{d\ln\gamma_R} \right]$$

$$\eta(\gamma_R) = \frac{-\left.\tau_{z\theta}\right|_{r=R}}{\dot{\gamma}_R}$$

Correction required





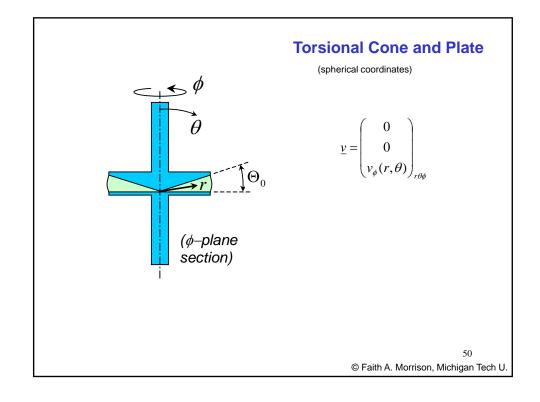
#### Torsional Parallel-Plate Flow - Normal Stresses

Similar tactics, logic (see Macosko, p221)

$$\left| \left( N_1 - N_2 \right) \right|_{\dot{\gamma}_R} = \frac{F_z}{\pi R^2} \left[ 2 + \frac{d \ln(F_z)}{d \ln \dot{\gamma}_R} \right]$$

(Not a direct material funcion)

Macosko, Rheology: *Principles, Measurements, and Applications*, VCH 1994, p220



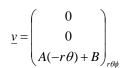
### To calculate shear rate:

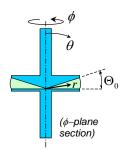
$$v_{\phi} = A(-r\theta) + B$$

$$v_{_{\phi}} = \frac{r\Omega}{\Theta_{0}} \bigg( \frac{\pi}{2} - \theta \bigg) \hspace{1cm} \text{(due to boundary conditions)}$$

$$\dot{\gamma} = \left| \dot{\underline{\gamma}} \right| = \left| \begin{pmatrix} 0 & 0 & r \frac{\partial}{\partial r} \begin{pmatrix} v_{\phi} \\ r \end{pmatrix} \\ 0 & 0 & \frac{\sin \theta}{r} \frac{\partial}{\partial \theta} \begin{pmatrix} v_{\phi} \\ \sin \theta \end{pmatrix} \\ r \frac{\partial}{\partial r} \begin{pmatrix} v_{\phi} \\ r \end{pmatrix} & \frac{\sin \theta}{r} \frac{\partial}{\partial \theta} \begin{pmatrix} v_{\phi} \\ \sin \theta \end{pmatrix} & 0 \end{pmatrix} \right|_{r\theta\phi}$$

$$\dot{\nu} = 2$$





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Result:

$$\underline{v} = \begin{pmatrix} 0 \\ 0 \\ \frac{r\Omega}{\Theta_0} \left( \frac{\pi}{2} - \theta \right) \end{pmatrix}_{r\theta\phi} \qquad \dot{\gamma} = \frac{\Omega}{\Theta_0} = \text{ constant}$$



Note: The shear rate is a constant.

The extra stresses  $\tau_{ij}$  are only a function of the shear rate, thus the  $\tau_{ii}$  are constant as well.

$$\underline{\underline{\tau}} = \begin{pmatrix} \tau_{rr} & 0 & 0 \\ 0 & \tau_{\theta\theta} & \tau_{\theta\phi} \\ 0 & \tau_{\phi\theta} & \tau_{\phi\phi} \end{pmatrix}_{r\theta\phi}$$

(viscometric flow)

Result:

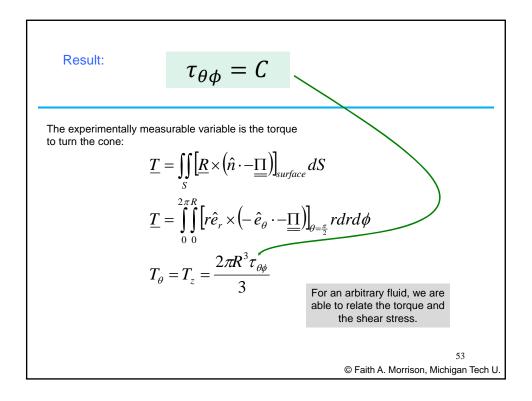
$$au_{ij} = ext{constant}$$

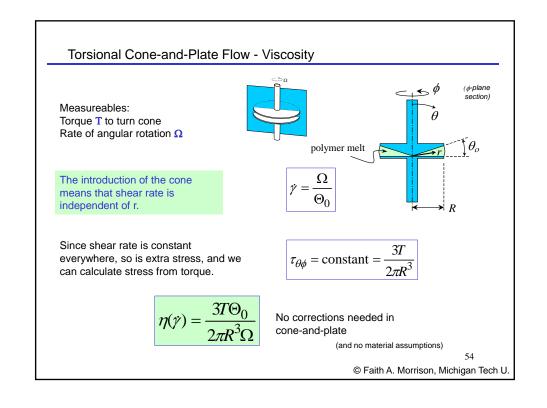
Torsional cone and plate is a homogeneous shear flow.

Assume:

- •Form of velocity
- •no φ-dependence
- •no slip
- •isothermal

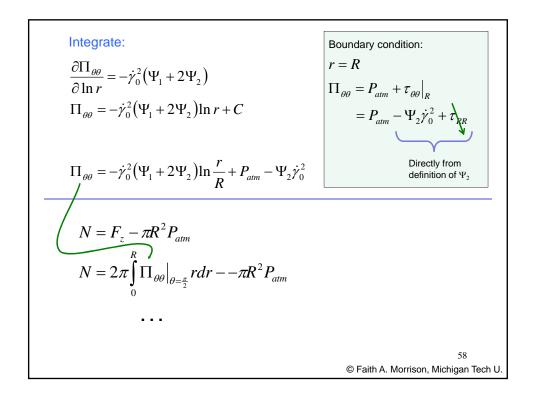
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On the bottom plate, 
$$sin\theta$$
=1,  $cos\theta$ =0: 
$$\begin{pmatrix} 0 \\ 0 \\ 0 \\ 0 \end{pmatrix}_{r\theta\phi} = \begin{pmatrix} -\frac{\partial P}{\partial r} \\ -\frac{1}{r}\frac{\partial P}{\partial \theta} \\ 0 \end{pmatrix}_{r\theta\phi} - \begin{pmatrix} \frac{2\tau_{rr}}{r} - \frac{\tau_{\theta\theta} + \tau_{\phi\phi}}{r} \\ 0 \\ 0 \end{pmatrix}_{r\theta\phi} = \begin{pmatrix} 0 \\ 0 \end{pmatrix}_{r\theta\phi} + \frac{2\tau_{rr}}{r} + \frac{\tau_{\theta\theta} + \tau_{\phi\phi}}{r} \\ 0 = -\frac{\partial (P + \tau_{\theta\theta})}{\partial r} - \frac{2\tau_{rr}}{r} + \frac{\tau_{\theta\theta} + \tau_{\phi\phi}}{r} \qquad \text{(valid to insert, since extra stress is constant)} \\ -\Psi_1 \dot{\gamma}_0^2 = \tau_{\phi\phi} - \tau_{\theta\theta} \\ -\Psi_2 \dot{\gamma}_0^2 = \tau_{\theta\theta} - \tau_{rr} \end{pmatrix} \qquad \text{(by definition)}$$
 
$$\frac{\partial \Pi_{\theta\theta}}{\partial \ln r} = -\dot{\gamma}_0^2 \left( \Psi_1 + 2\Psi_2 \right)$$

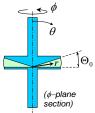
Result: 
$$\frac{\partial \Pi_{\theta\theta}}{\partial \ln r} = -\dot{\gamma}_0^2 \big( \Psi_1 + 2 \Psi_2 \big)$$
 The experimentally measurable variable is the fluid thrust on the plate minus the thrust of  $P_{atm}$ : 
$$N = F_z - \pi R^2 P_{atm}$$
 
$$\underline{F} = \iint_S \big[ \big( \hat{n} \cdot -\underline{\underline{\Pi}} \big) \big]_{surface} dS$$
 
$$\underline{F} = \int_0^2 \int_0^R \big[ \big( -\hat{e}_\theta \cdot -\underline{\underline{\Pi}} \big) \big]_{\theta = \frac{\pi}{2}} r dr d\phi$$
 
$$F_\theta = F_z = 2\pi \int_0^R \Pi_{\theta\theta} \big|_{\theta = \frac{\pi}{2}} r dr$$
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#### Torsional Cone-and-Plate Flow – 1st Normal Stress

Measureables: Normal thrust **F** 





$$N = \left[ 2\pi \int_{0}^{R} \Pi_{\theta\theta} \Big|_{\theta = \frac{\pi}{2}} r dr \right] - \pi R^{2} p_{atm}$$

The total upward thrust of the cone can be related directly to the first normal stress coefficient.

$$\Psi_1(\dot{\gamma}) = \frac{2F\Theta_0^2}{\pi R^2 \Omega^2}$$

(see also DPL pp522)

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### Torsional Cone-and-Plate Flow – 2<sup>nd</sup> Normal Stress

 $\Pi_{\theta\theta} = -\dot{\gamma}_0^2 \left( \Psi_1 + 2\Psi_2 \right) \ln \frac{r}{R} + P_{atm} - \Psi_2 \dot{\gamma}_0^2$ 

If we obtain  $\Pi_{\theta\theta}$  as a function of r / R, we can also obtain  $\Psi_2$ .

•MEMS used to manufacture sensors at different radial positions

The Normal Stress Sensor System (NSS)





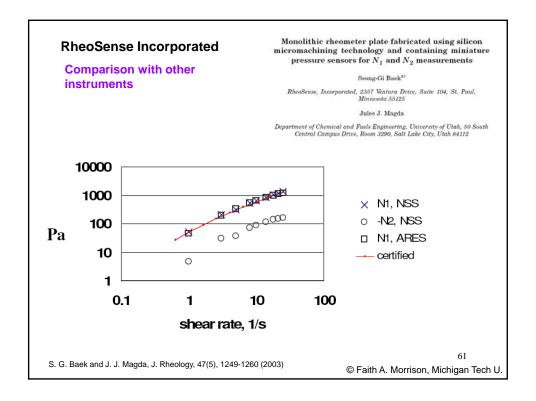


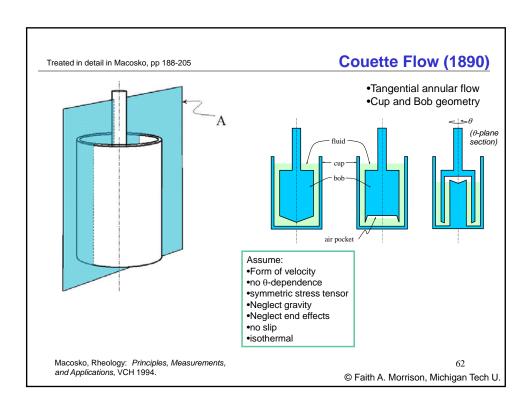
Patented Technology

RheoSense Incorporated (www.rheosense.com)

- •S. G. Baek and J. J. Magda, J. Rheology, 47(5), 1249-1260 (2003)
- •J. Magda et al. Proc. XIV International Congress on Rheology, Seoul, 2004.

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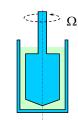
#### **Couette Flow**

Assume:

- •Form of velocity
- •no θ-dependence
- •symmetric stress tensor
- Neglect gravity
- •Neglect end effects
- •no slip
- •isothermal

$$\eta = \frac{T(\kappa - 1)}{2\pi R^2 L \kappa^3 \Omega}$$

$$\kappa = \frac{R_{inner}}{R}$$



•Generates a lot of signal

•Is widely available

•Is well understood

•Can go to high shear rates

BUT

As with many measurement systems, the assumptions made in the analysis do not always hold:

- •End effects are not negligible
- •Wall slip occurs with many systems
- •Inertia is not always negligible
- •Secondary flows occur (cup turning is more stable than bob turning to inertial instabilities; there are elastic instabilities; there are viscous heating instabilities)
- Alignment is important
- Viscous heating occurs
- •Methods for measuring  $\Psi_1$  are error prone
- •Cannot measure Ψ<sub>2</sub>

Macosko, Rheology: Principles, Measurements, and Applications, VCH 1994.

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For the PP and CP geometries, we can also calculate G', G".

Parallel plate 
$$\eta'(\omega) = \frac{G''(\omega)}{\omega} = \frac{2HT_0 \sin \delta}{\pi R^4 \omega \theta_0}$$
 
$$\eta''(\omega) = \frac{G'(\omega)}{\omega} = \frac{2HT_0 \cos \delta}{\pi R^4 \omega \theta_0}$$
 Amplitude of oscillation

Cone and plate 
$$\eta'(\omega) = \frac{3\Theta_0 T_0 \sin \delta}{2\pi R^3 \omega \phi_0}$$
 
$$\eta''(\omega) = \frac{3\Theta_0 T_0 \cos \delta}{2\pi R^3 \omega \phi_0}$$
 Amplitude of accillation

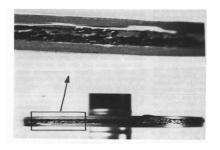
$$\eta''(\omega) = \frac{3\Theta_0 I_0 \cos \delta}{2\pi R^3 \omega \phi_0}$$

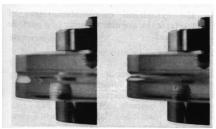
oscillation

- •A typical diameter is between 8 and 25mm; 30-40mm are also used
- •To increase accuracy, larger plates (R larger) are used for less viscous materials to generate more torque.
- •Amplitude may also be increased to increase torque
- •A complete analysis of SAOS in the Couette geometry is given in Sections 8.4.2-3

# Limits on Measurements: Flow instabilities in rheology

# Cone and plate/Parallel plate flow





Figures 6.7 and 6.8, p. 175 Hutton; PDMS

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### Limits on Measurements: Flow instabilities in rheology

### **Taylor-Couette flow**

1923 GI Taylor; inertial instability 1990 Ron Larson, Eric Shaqfeh, Susan Muller; elastic instability



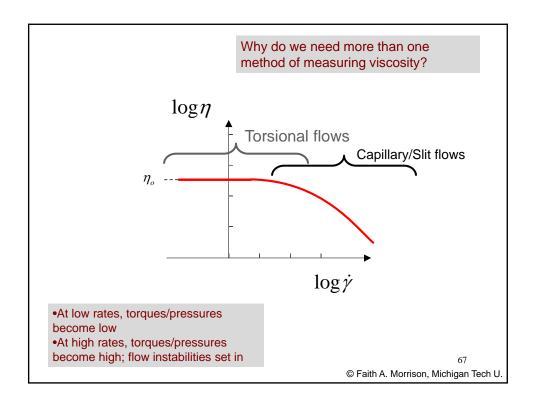




FIGURE 11. Flow visualizations in a Taylor-Couette cell. (a) Newtonian fluid at high Taylor number (Ta = 3800); (b) Boger fluid at negligible Taylor number ( $Ta = 9.6 \times 10^{-9}$ ) shortly after the onset of secondary flow ( $t_a$  in figure 9); (c) Boger fluid at negligible Taylor number after full development of secondary flow ( $t_a$  in figure 9).

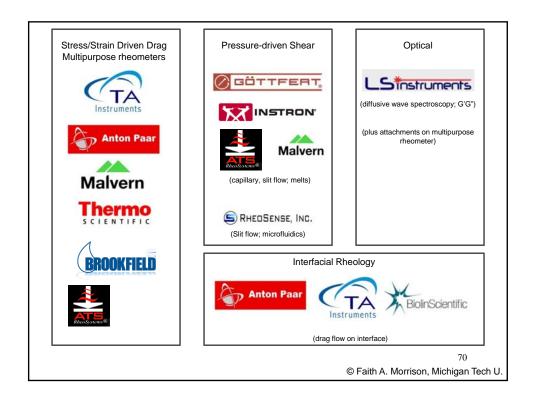
•GI Taylor "Stability of a viscous liquid contained between two rotating cylinders," *Phil. Trans. R. Soc. Lond.* A 223, 289 (1923) •Larson, Shaqfeh, Muller, "A purely elastic instability in Taylor-Couette flow," *J. Fluid Mech*, 218, 573 (1990)

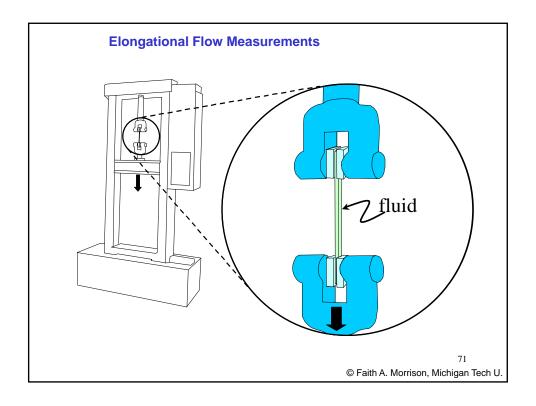
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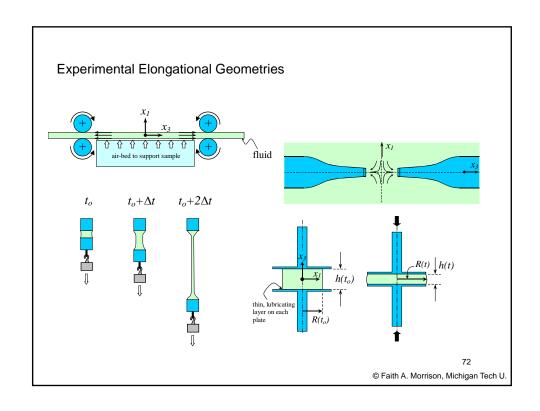


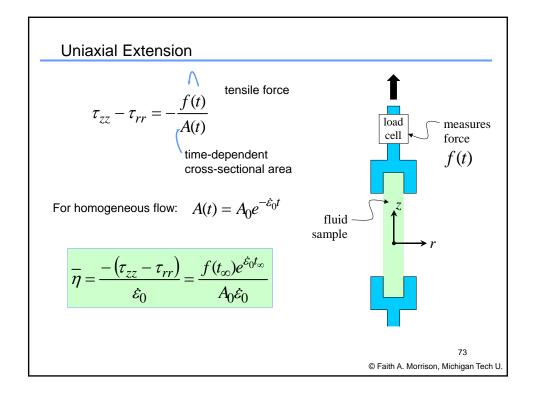
$L = \text{capillary length}$ $\mathcal{R} = \frac{1}{4} \left[ 3 + \frac{d \ln(4Q)\pi R^3}{2 \ln \tau_8} \right]$ $\tau_R = \tau_{r_T} t_{l-R}$ $Farallel disk (at tim)$ $\mathcal{T} = \text{torque on top plate}$ $\Omega = \text{angular velocity of top plate}, > 0$ $\mathcal{H} = \text{gap}$ $\mathcal{R} = \frac{1}{4} \left[ 3 + \frac{d \ln(\mathcal{T}/2\pi R^3)}{d \ln \pi^2} \right]$ $\dot{\gamma}_R = \dot{\gamma}(R)$ $Cone \text{ and plate}$ $\mathcal{T} = \text{torque on plate}$ $\mathcal{D} = \text{angular velocity of cone}, > 0$	Function $\eta = \frac{\tau_R}{4Q/\pi R^3} \mathcal{R}$
Capillary flow (wall conditions) $\mathcal{P}_0, \mathcal{P}_L = \text{modified pressure at } z = 0, L$ $\mathcal{Q} = \text{flow rate}$ $\mathcal{Q} = \text{flow rate}$ $\mathcal{L} = \text{capillary legals}$ $\mathcal{R} = \frac{1}{4} \left[ \frac{1}{4} + \frac{4 \ln (40/rR^2)}{4 \ln r_0} \right]$ $\tau_F = \nabla \tau_{C_F} - R$ Parallel disk (at rim) $\mathcal{T} = \text{torque on top plate}$ $\Omega = \text{angular velocity of top plate}, > 0$ $\mathcal{H} = \text{gap}$ $\mathcal{H} = \frac{1}{4} \left[ \frac{1}{3} + \frac{4 \ln (7/2\pi R^2)}{\ln r_F} \right]$ $\dot{\gamma}_R = \dot{\gamma}(R)$ Cone and plate $\mathcal{T} = \text{torque on plate}$	$\eta = \frac{2T^\prime}{\piR^3\dot{\gamma}_R}\mathcal{F}$
$\begin{array}{c} \mathcal{P}_0, \mathcal{P}_L = \operatorname{modified pressure at } z = 0, L & \frac{(\mathcal{P}_0 - \mathcal{P}_L)R}{2L} & \frac{4Q}{\pi R^3} \mathcal{R} \\ Q = \operatorname{flow rate} \\ L = \operatorname{capillary length} \\ \mathcal{R} = \frac{1}{4} \left[ 3 + \frac{4 \ln(4Q) r R^3}{4 \ln r x^2} \right] \\ v_R = v_{r_L}  _{r=R} \end{array}$ $\begin{array}{c} Parallel disk (fat tim) \\ \mathcal{T} = \operatorname{torque} \text{on top plate} \\ \Omega = \operatorname{angular velocity of top plate}, > 0 & \frac{2\mathcal{T}}{\pi R^3} \mathcal{R} & \frac{r\Omega}{H} \\ H = \operatorname{gap} \\ \mathcal{R} = \frac{1}{4} \left[ 3 + \frac{4 \ln(T/2 r R^3)}{4 \ln r x} \right] \\ \dot{y}_R = \dot{y}(R) \\ \dot{z}_R = \dot{y}(R) \end{array}$ $\begin{array}{c} Cone \text{ and plate} \\ \mathcal{T} = \operatorname{torque on plate} \\ \mathcal{T} = \operatorname{torque on plate} \\ \mathcal{T} = \operatorname{torque on plate} \\ \mathcal{D} = \operatorname{angular velocity of cone}, > 0 \end{array}$	$\eta = \frac{2T^\prime}{\piR^3\dot{\gamma}_R}\mathcal{F}$
$L = \text{capillary length}$ $\mathcal{R} = \frac{1}{4} \left[ 3 + \frac{d \ln(4\Omega)\pi R^3}{d \ln \tau_R} \right]$ $\tau_R = \tau_{r_T}  _{r=R}$ $Farallel disk (at tim)$ $\mathcal{T} = \text{torque on top plate}$ $\Omega = \text{angular velocity of top plate}, > 0$ $\mathcal{H} = \text{gap}$ $\mathcal{R} = \frac{1}{4} \left[ 3 + \frac{d \ln(T/2\pi R^3)}{d \ln T^2} \right]$ $\dot{\gamma}_R = \dot{\gamma}_R R^3$ $\mathcal{C}$ $Cone and plate \mathcal{T} = \text{torque on plate} \mathcal{D} = \text{angular velocity of cone}, > 0$	$\eta = \frac{2T^\prime}{\piR^3\dot{\gamma}_R}\mathcal{F}$
$\mathcal{R} = \frac{1}{4} \left\{ 3 + \frac{4 \ln(4 \otimes (nR^2)}{d \log x} \right\}$ $\tau_R = \tau_{r_L} _{r=R}$ $Parallel disk (at rim)$ $\mathcal{T} = torque on top plate$ $\Omega = angular velocity of top plate, > 0$ $\mathcal{H} = gap$ $\mathcal{R} = \frac{1}{4} \left\{ 3 + \frac{4 \ln(T/2\pi R^2)}{d \ln T_R} \right\}$ $\dot{\gamma}_R = \dot{\gamma}(R)$ $\dot{\gamma}_R = \dot{\gamma}(R)$ $Cone and plate \mathcal{T} = torque on plate \mathcal{T} = torque on plate \mathcal{T} = thrust on plate \mathcal{T} = thrust on plate \Omega = angular velocity of cone, > 0$	
$\tau_R = \tau_{r_1} _{r=R}$ $Parallel disk (at rim)$ $\mathcal{T} = torque on top plate$ $\Omega = angust velocity of top plate, > 0$ $\mathcal{H} = gap$ $\mathcal{R} = \frac{1}{3} + \frac{4m(\mathcal{T}/2\pi^2)}{4n^2\pi}$ $\dot{\gamma}_R = \dot{\gamma}(R)$ $Cone and plate$ $\mathcal{T} = torque on plate$ $\mathcal{T} = torque on plate$ $\mathcal{F} = thrust on plate$ $\Omega = angustar velocity of cone, > 0$	
$\tau_R = \tau_{r_1} _{r=R}$ $Parallel disk (at rim)$ $\mathcal{T} = torque on top plate$ $\Omega = angust velocity of top plate, > 0$ $\mathcal{H} = gap$ $\mathcal{R} = \frac{1}{3} + \frac{4m(\mathcal{T}/2\pi^2)}{4n^2\pi}$ $\dot{\gamma}_R = \dot{\gamma}(R)$ $Cone and plate$ $\mathcal{T} = torque on plate$ $\mathcal{T} = torque on plate$ $\mathcal{F} = thrust on plate$ $\Omega = angustar velocity of cone, > 0$	
$T$ = torque on top plate $\Omega$ = angular velocity of top plate, $> 0$ $\frac{2T}{\pi R^3}R$ $\frac{r\Omega}{H}$ $H$ = gap $R$ = $\frac{1}{4}\left[3 + \frac{d\ln{(T/2\pi R^3)}}{d\ln{r_2}}\right]$ $\dot{\gamma}_R = \dot{\gamma}(R)$ Cone and plate $T$ = torque on plate $T$ = torque on plate $T$ = thrust on plate $T$ = thrust on plate $T$ =	
$Ω$ = angular velocity of top plate, $> 0$ $\frac{1}{\pi R^2}R$ $\frac{1}{R}$ $H$ = gap $R$ = $\frac{1}{4}\left[3 + \frac{d \ln{(T/2\pi R^2)}}{d \ln{T^2}}\right]$ $\dot{γ}_R = \dot{γ}(R)$ Cone and plate $T$ = toeque on plate $T$ = toeque on plate $T$ = thrust on plate $T$ = thrust on plate $T$ = thrust on plate $T$ = $T$	
$H = \sup_{\mathcal{R}} H = \left\{ \left[ 3 + \frac{d \ln \left( T / 2 \pi R^3 \right)}{d \ln \gamma_R} \right] \right.$ $\dot{\gamma}_R = \dot{\gamma} \left( R \right)$ $Cone \ and \ plate$ $\mathcal{T} = \ torque \ on \ plate$ $\mathcal{F} = \ threst \ on \ plate$ $\mathcal{F} = \ threst \ on \ plate$ $\Omega = \ angular \ velocity \ of \ cone, \ > 0$	
$\mathcal{R} = \frac{1}{4} \left[ 3 + \frac{4 \ln (\mathcal{T} f \mathcal{R} R^2)}{d \ln \gamma_R} \right]$ $\dot{\gamma}_R = \mathcal{V}(R)$ Cone and plate $\mathcal{T} = \text{torque on plate}$ $\mathcal{F} = \text{thrust on plate}$ $\Omega = \text{angular velocity of cone, } > 0$	3T \(\theta_0\)
$\dot{\gamma}_R = \dot{\gamma}(R)$ Cone and plate $\mathcal{T} = \text{torque on plate}$ $\mathcal{F} = \text{thrust on plate}$ $\Omega = \text{angular velocity of cone, } > 0$	3T\Theta_0
Cone and plate $\mathcal{T} = \text{torque on plate}$ $\mathcal{T} = \text{thrus to n plate}$ $\mathcal{T} = \text{thrus to n plate}$ $\Omega = \text{angular velocity of cone, } > 0$	3T\O_0
$\mathcal{T}=$ torque on plate $\mathcal{T}=$ thrust on plate $\mathcal{F}=$ thrust on plate $\Omega=$ angular velocity of cone, $>0$	3700
$\mathcal{F}=$ thrust on plate $\dfrac{2\piR^3}{\Theta_0}$ $\Theta_0$ $\Omega=$ angular velocity of cone, $>0$	3 T ⊕0
$\Omega$ = angular velocity of cone, > 0	
$\Theta_0 = \text{cone angle}$	$\Psi_1 = \frac{2\mathcal{F}\Theta_0^2}{\pi R^2 \Omega^2}$
Couette (bob turning)	
$T' = \text{torque on inner cylinder, } < 0$ $\frac{-T'}{2\pi R^2 L \kappa^2} \frac{\kappa \Omega}{1 - \kappa}$ $\Omega = \text{angular velocity of bob, } > 0$ $\frac{-T'}{2\pi R^2 L \kappa^2} \frac{\kappa \Omega}{1 - \kappa}$	$\eta = \frac{T'(\kappa - 1)}{2\pi R^2 L \kappa^3}$
and the same of th	$^{1/2} = 2\pi R^2 L \kappa^3$
R = outer radius	
$\kappa R = \text{inner radius}$ $L = \text{length of bob}$	
L = length of bob	
Couette (cup turning)	
$T'$ = torque on inner cylinder, > 0 $\frac{T}{2\pi R^2 L \kappa^2} = \frac{\kappa \Omega}{1 - \kappa}$ $\Omega = \text{angular velocity of cup, > 0}$ $\frac{T}{2\pi R^2 L \kappa^2} = \frac{\kappa \Omega}{1 - \kappa}$	$\eta = \frac{T'(1-\kappa)}{2\pi R^2 L \kappa^3}$
and the second s	$\eta = \frac{1}{2\pi R^2 I_{-K}^3 s}$
R = outer radius  ISO Macosko, Part II $\kappa R = \text{inner radius}$	2011 201
	2011 201

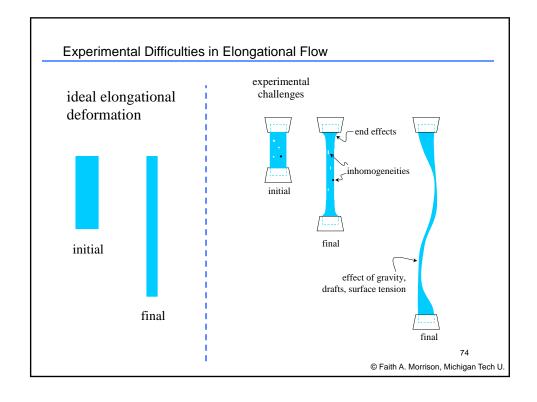
#### Shear measurements Pros and Cons Comparison of Experimental Features of Four Common Shear Geometries Feature Parallel Disk Cone and Plate Capillary Couette (Cup and Bob) Good for high viscosity Good for high viscosity Good for high viscosities Good for low viscosities Melt fracture at very high rates, i.e., distorted extrudates and pressure fluctuations are observed are high De Taylor cells are observed at high De Edge fracture at modest rates Flow stability Edge fracture at modest rates Sample size and sample < 1 g; easy to load loading < 1 g; highly viscous materials can be difficult to load 10-20 g; highly viscous materials can be difficult to load 40 g minimum; easy to load Correction on shear rate needs to be applied; this correction is ignored in most commercial software packages Multiple corrections need to be Straightforward applied Data handling Straightforward No; shear rate and shear stress vary with radius No; shear rate and shear stress vary with radius Homogeneous? Yes (small core angles) Yes (narrow gap) High pressures in reservoir cause problems with compressibility of melt Pressure effects Maximum shear rate is limited by edge fracture; usually cannot obtain shear-thinning data Maximum shear rate is limited by sample leaving cup due to either inertia or elastic effects; also 3-D secondary flows develop Shear rates Maximum shear rate is limited Very high rates accessible by edge fracture; usually cannot obtain shear-thinning data Good for stiff samples, even gels; wide range of temperatures possible $\Psi_1$ measurable; wide range of temperatures possible Constant-Q or constant- $\Delta \mathcal{P}$ modes available; wide range of temperatures possible © Faith A. Morrison, Michigan Tech U.

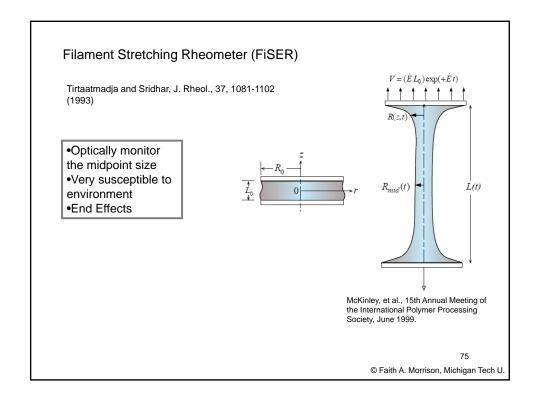










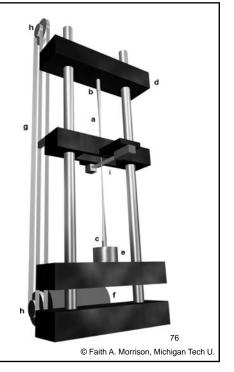


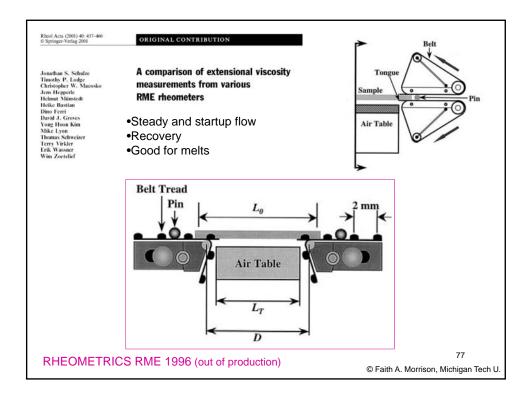
### Filament Stretching Rheometer

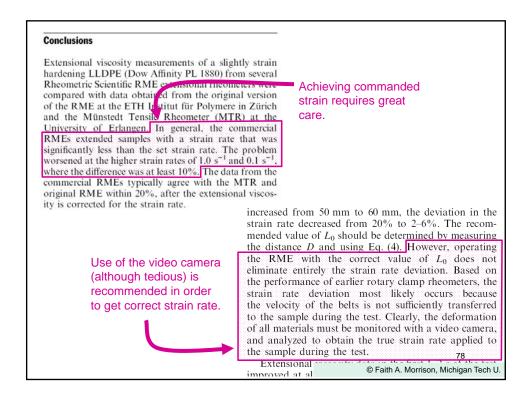
(Design based on Tirtaatmadja and Sridhar)

"The test sample (a) undergoing investigation is placed between two parallel, circular discs (b) and (c) with diameter 2R<sub>0</sub>=9 mm. The upper disc is attached to a movable sled (d), while the lower disc is in contact with a weight cell (e). The upper sled is driven by a motor (f), which also drives a mid-sled placed between the upper sled and the weight cell; two timing belts (g) are used for transferring momentum from the motor to the sleds. The two toothed wheels (h), driving the timing belts have a 1:2 diameter ratio, ensuring that the mid-sled always drives at half the speed of the upper sled. This means that if the mid-sled is placed in the middle between the upper and the lower disc at the beginning of an experiment, it will always stay midway between the discs. On the mid-sled, a laser (i) is placed for measuring the diameter of the mid-filament at all times.

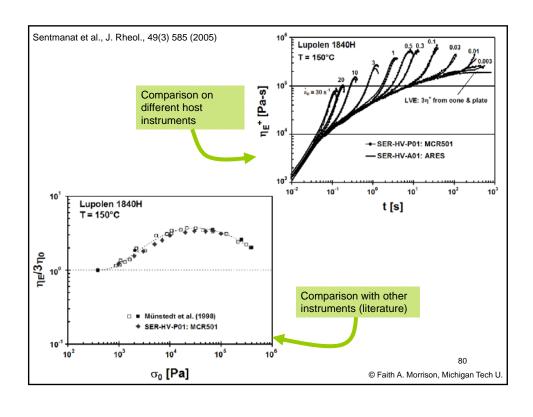
Bach, Rasmussen, Longin, Hassager, JNNFM 108, 163 (2002)



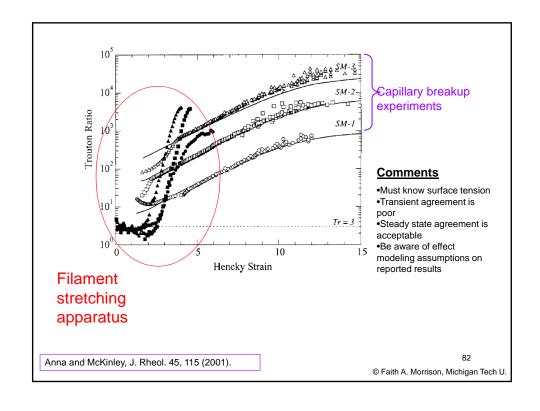


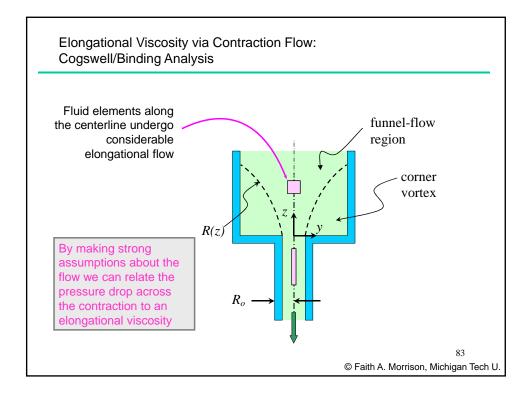


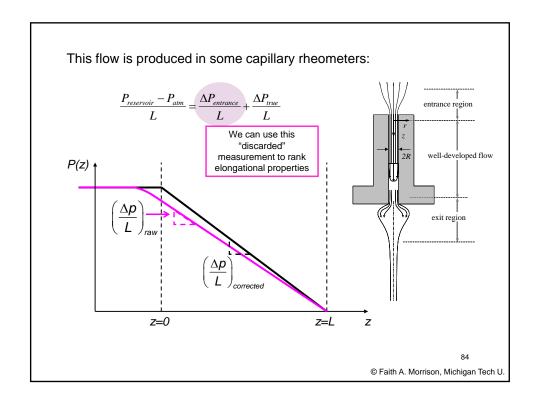
# Sentmanat Extension Rheometer (2005) •Originally developed for rubbers, good for melts •Measures elongational viscosity, startup, other material functions •Two counter-rotating drums •Easy to load; reproducible $\epsilon_{\rm H} = 1.000$ www.xpansioninstruments.com http://www.xpansioninstruments.com/rheo-optics.htm 79 © Faith A. Morrison, Michigan Tech U.

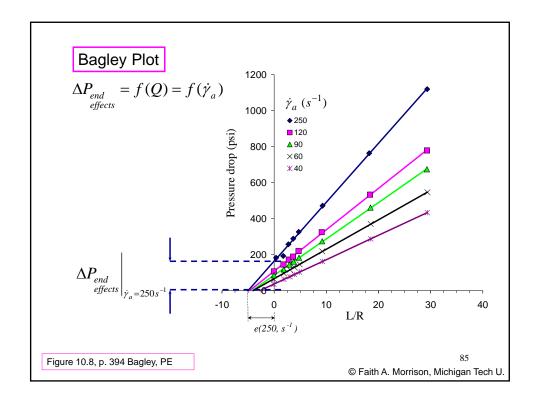


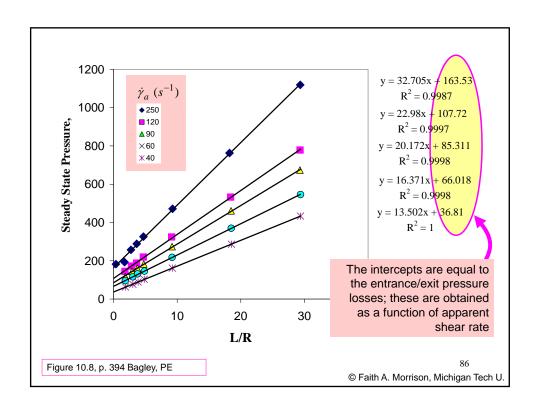
# CaBER Extensional Rheometer •Polymer solutions •Works on the principle of capillary filament break up •Cambridge Polymer Group and HAAKE For more on theory see: campoly.com/notes/007.pdf Brochure: www.thermo.com/com/cda/product/detail/1,,17848,00.html Operation •Impose a rapid step elongation •form a fluid filament, which continues to deform •flow driven by surface tension •also affected by viscosity, elasticity, and mass transfer •measure midpoint diameter as a function of time •Use force balance on filament to back out an apparent elongational viscosity





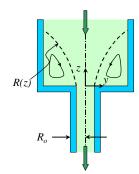






### **Assumptions for the Cogswell Analysis**

- incompressible fluid
- funnel-shaped flow; no-slip on funnel surface
- unidirectional flow in the funnel region
- well developed flow upstream and downstream
- θ-symmetry
- pressure drops due to shear and elongation may be calculated separately and summed to give the total entrance pressure-loss
- neglect Weissenberg-Rabinowitsch correction
- shear stress is related to shear-rate through a power-law
- elongational viscosity is constant
- shape of the funnel is determined by the minimum generated pressure drop
- no effect of elasticity (shear normal stresses neglected)
- neglect inertia



$$\dot{\gamma} \approx \dot{\gamma}_a$$

$$\underline{\tau_R} = m\dot{\gamma}_a^n$$

$$\underline{\eta} = \text{constant}$$

F. N. Cogswell, Polym. Eng. Sci. (1972) 12, 64-73. F. N. Cogswell, Trans. Soc. Rheol. (1972) 36, 383-403.

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# **Cogswell Analysis**

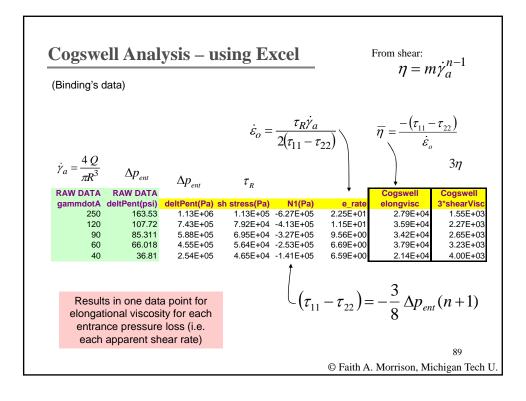
$$\dot{\varepsilon}_o = \frac{\tau_R \dot{\gamma}_a}{2(\tau_{11} - \tau_{22})}$$

$$\dot{\varepsilon}_o = \frac{\tau_R \dot{\gamma}_a}{2(\tau_{11} - \tau_{22})} \qquad \begin{aligned} \tau_R &= \eta \, \dot{\gamma}_a \\ \dot{\gamma}_a &= \frac{4Q}{\pi P^3} \end{aligned} \qquad (\eta = m \dot{\gamma}_a^{n-1})$$

elongation normal

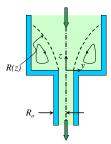
stress 
$$(\tau_{11} - \tau_{22}) = -\frac{3}{8} \Delta p_{ent}$$
  $(n+1)$ 

elongation viscosity 
$$\overline{\eta} \approx \frac{-(\tau_{11} - \tau_{22})}{\dot{\varepsilon}_o} = \frac{\frac{9}{32}(n+1)^2 \Delta p_{ent}^2}{\tau_R \dot{\gamma}_a}$$



# Assumptions for the Binding Analysis

- incompressible fluid
- funnel-shaped flow; no-slip on funnel surface
- unidirectional flow in the funnel region
- •well developed flow upstream and downstream
- θ -symmetry
- shear viscosity is related to shear-rate through a
- elongational viscosity is given by a power law
- shape of the funnel is determined by the minimum work to drive flow
- no effect of elasticity (shear normal stresses neglected)
- the quantities  $(dR/dz)^2$  and  $d^2R/dz^2$ , related to the shape of the funnel, are neglected; implies that the radial velocity is neglected when calculating the rate of deformation
- neglect energy required to maintain the corner circulation
- neglect inertia



$$\frac{\tau_R = m\dot{\gamma}_a^n}{\eta = l\dot{\varepsilon}_o^{t-1}}$$

D. M. Binding, JNNFM (1988)

# **Binding Analysis**

- I, elongational prefactor

$$\Delta p_{ent} = \frac{2m(1+t)^2}{3t^2(1+n)^2} \left\{ \frac{t(3n+1)n^t I_{nt}}{m} \right\}^{1/(1+t)} \dot{\gamma}_{R_o}^{t(n+1)/(1+t)} \left\{ 1 - \alpha^{3t(n+1)/(1+t)} \right\}$$

$$I_{nt} = \int_{0}^{1} \left| 2 - \left( \frac{3n+1}{n} \right) \phi^{1+1/n} \right|^{t+1} \phi \, d\phi$$

$$\dot{\gamma}_{R_o} = \frac{(3n+1)}{n \pi R_o^3} Q$$

elongation viscosity 
$$\eta = l\dot{\varepsilon}_o^{t-1}$$

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# **Binding Analysis**

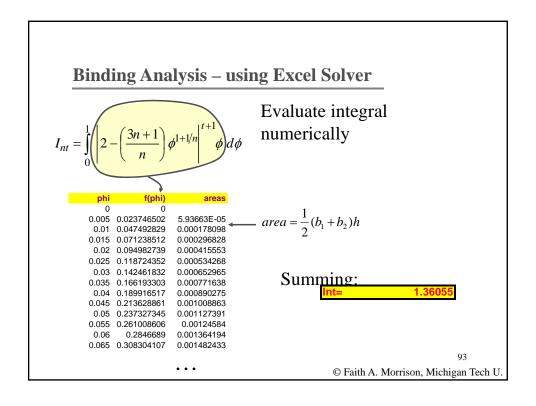
Note: there is a non-iterative solution method described in the text; The method using Solver is preferable, since it uses all the data in finding optimal values of 1 and t.

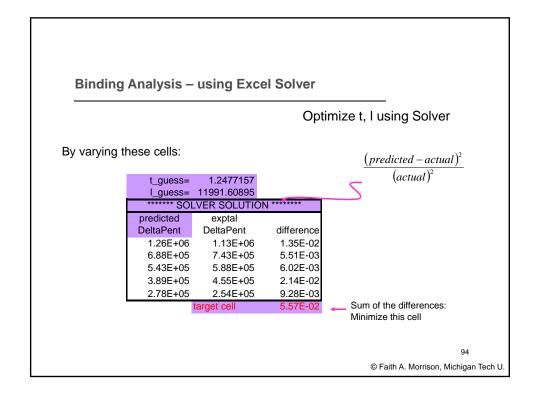
## **Evaluation Procedure**

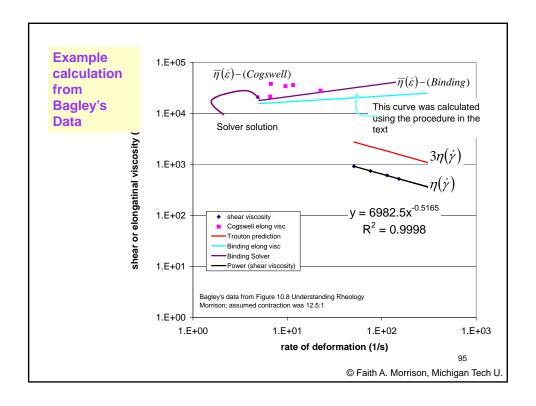
- 1. Shear power-law parameter n must be known; must have data for  $\Delta p_{ent}$  versus Q
- 2. Guess t, 1
- 3. Evaluate  $I_{nt}$  by numerical integration over  $\phi$
- 4. Using Solver, find the best values of t and l that are consistent with the  $\Delta p_{ent}$  versus Q data

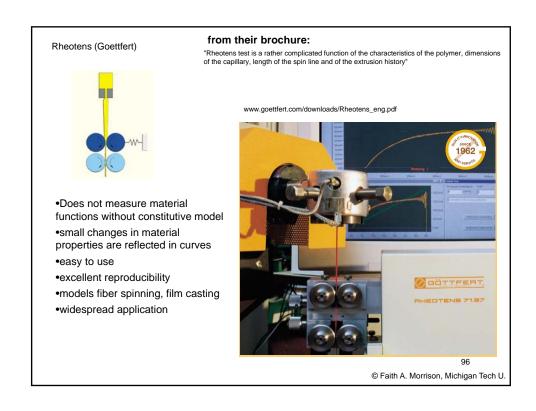
Results in values of t, I for a model (power-law)

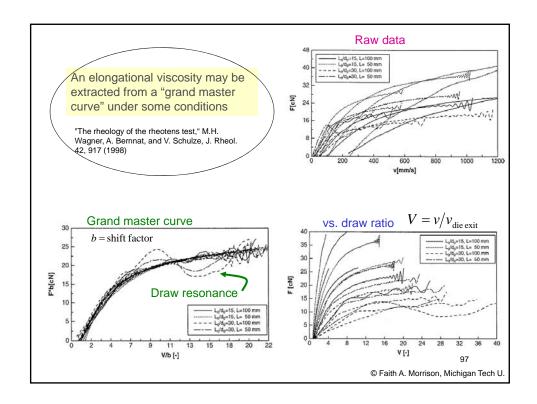
92



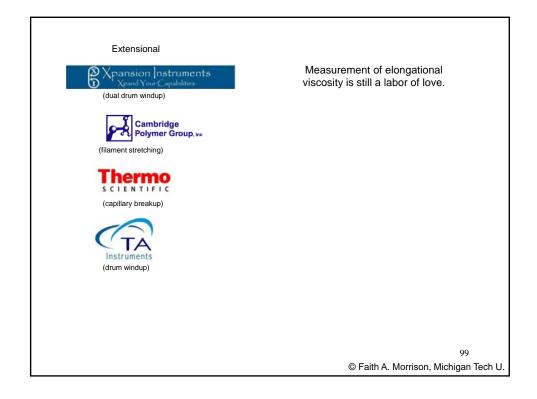


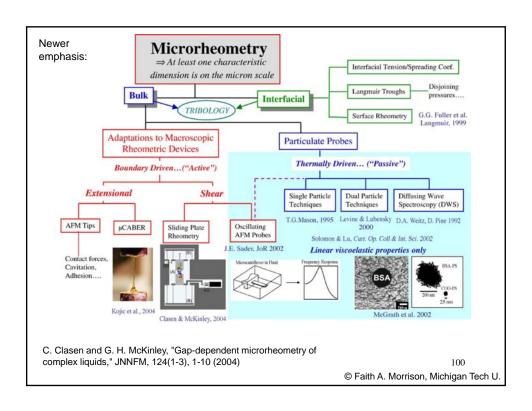


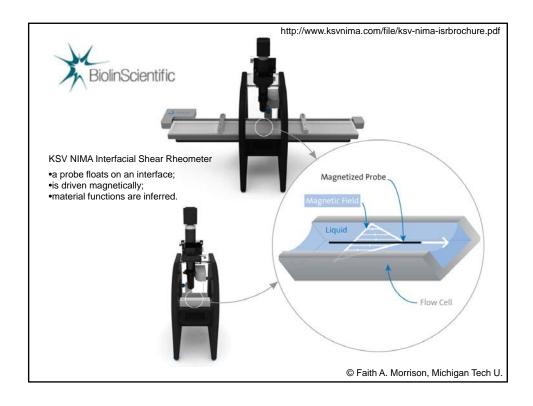




Elongational measurements	-			Filament	Binding/
	Feature	Melt Stretching	MBER	Stretching	Cogswell
Pros and Cons	Stress Range	Good for high viscosity	Good for high viscosity	Good for low viscosity at room temperature	Good for high and low viscosities
	Flow stability	Subject to gravity, surface tension and air currents	Can be unstable at high rates	Subject to gravity, surface tension and air currents	Unstable at very high rates
	Sample size and sample loading	10 g; care must be taken to minimize end effects	<2 g; requires careful preparation and loading	<1 g; easy to load	40 g minimum; eas to load
	Data handling	Straightforward, but does not result in any elongational material functions	Straightforward; more involved if strain is measured	Two tests are required to account for strain inhomogeneities	Cogswell— straightforward Binding—more complicated but no difficult
	Homogeneous?	No, not at ends	Could be with care	No, not at ends	No-mixed shear and elongational flow
	Pressure effects	No	No	No	Yes— compressibility of melt reservoir could cause difficulties
	Elongation rates	Maximum rates depend on clamp speeds	Maximum elongation rate is limited by ability to maintain the sample in steady flow	Maximum rates depend on plate speeds; minimum rates depend on the ratio of gravity and viscous effects	High and low rates possible
	Special features	Cannot reach high strains or steady state; wide range of temperatures is possible; the instrument is commercially available	Often strain is not measured but is calculated from the imposed strain rate; a wide range of temperatures is possible; the instrument is commercially available	Currently limited to room temperature liquids	Is based on a presumed funnel- shaped flow—this may not take place; wide range of temperatures possible

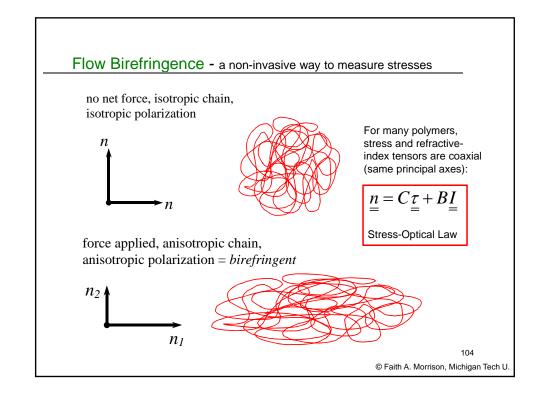


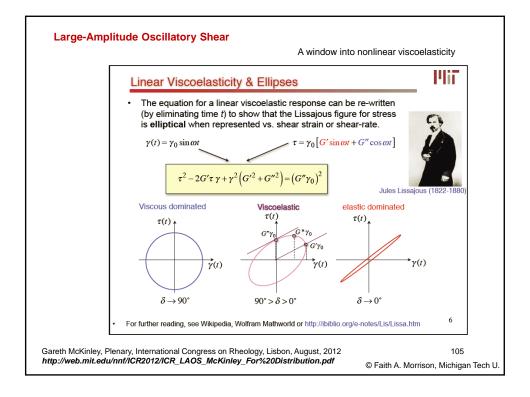












## **Summary**

### **SHEAR**

- •Shear measurements are readily made
- •Choice of shear geometry is driven by fluid properties, shear rates
- •Care must be taken with automated instruments (nonlinear response, instrument inertia, resonance, motor dynamics, modeling assumptions)
- •Microrheometry

## **ELONGATION**

- •Elongational properties are still not routine
- •Newer instruments (Sentmanat,CaBER) have improved the possibility of routine elongational flow measurements
- •Some measurements are best left to the researchers dedicated to them due to complexity (FiSER)
- •Industries that rely on elongational flow properties (fiber spinning, foods) have developed their own ranking tests

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# **Summary**

- 1. Introduction
- 2. Math review
- 3. Newtonian Fluids
- 4. Standard Flows
- 5. Material Functions
- 6. Experimental Data
- 7. Generalized Newtonian Fluids
- 8. Memory: Linear Viscoelastic Models
- 9. Advanced Constitutive Models
- 10. Rheometry

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