

**CE2201 Exam 3**

**Name** \_\_\_\_\_

April 26, 2010

2 Hours

Closed Book

Closed Notes

(3) 3x5 Note-cards Allowed

Calculators Allowed

No Collaboration

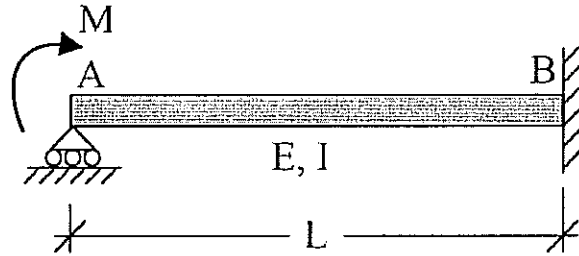
Read each question carefully. Provide the information required.

Show your calculations for credit.

If I cannot read it, I cannot grade it.

**Problem 1.** The following indeterminate beam is loaded by a concentrated moment at point A (30 points).

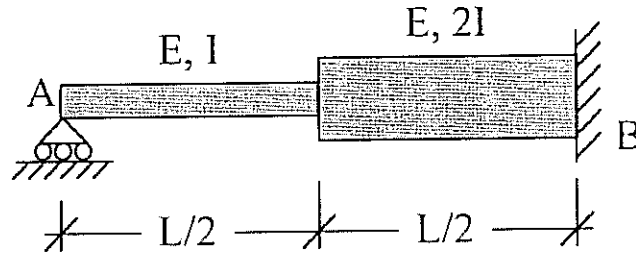
- Find the support reactions using the flexibility method.
  - o Clearly draw and identify your released structure.
  - o Identify your equation(s) of compatibility.
- Draw the shear and moment diagrams for this loading.



Ignore axial effects

**Problem 2.** The following indeterminate beam is not loaded, but the support at point A settles downward by a distance of  $\delta$ . The beam cross-section is *not* constant (20 points).

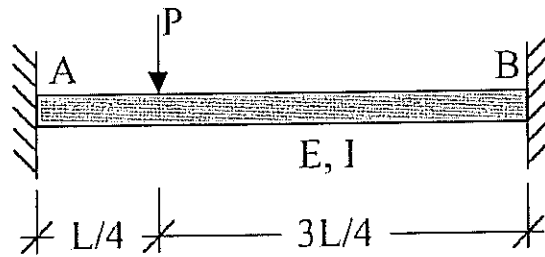
- Find the reaction at support A using the flexibility method.
  - o Clearly draw and identify your released structure.
  - o Identify your equation(s) of compatibility.
  - o Hint: note that  $\delta$  due to external loading is zero.



Ignore axial effects

**Problem 3.** The following indeterminate beam is loaded at mid-span (30 points).

- Clearly draw and identify a stable, determinate released structure.
- Derive the equation(s) of compatibility for your released structure in terms of the unknown, released reactions (You must find the appropriate deformations of the released structure. *Do not solve for the unknown reactions*).

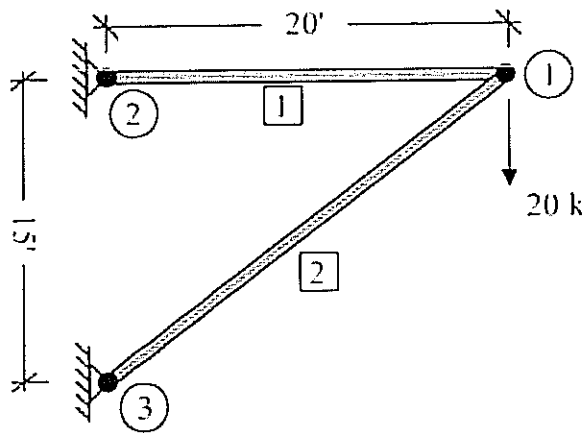


Ignore axial effects

**Problem 4.** The following indeterminate truss is loaded at joint 1. Both members have an Area of 0.5 in and a Modulus of Elasticity of 29,000 ksi (20 points).

- The local 4x4 stiffness matrices for use in the direct-stiffness method are given.
- Fill in the terms of the global stiffness matrix.
- Subdivide the system into free and supported degrees-of-freedom (i.e., identify  $K_{11}$  and  $K_{21}$  as well as  $Q_f$ ,  $\delta_f$ , and  $Q_s$ ).
- Find the x and y-deflections at the free joint using the direct-stiffness method.
- Find the support reactions using the direct-stiffness method.

$$K_1 = \begin{matrix} & \begin{matrix} \textcircled{1x} & \textcircled{1y} & \textcircled{2x} & \textcircled{2y} \end{matrix} \\ \begin{matrix} \textcircled{1x} \\ \textcircled{1y} \\ \textcircled{2x} \\ \textcircled{2y} \end{matrix} & \begin{bmatrix} 60.4 & 0 & -60.4 & 0 \\ 0 & 0 & 0 & 0 \\ -60.4 & 0 & 60.4 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \end{matrix} \quad K_2 = \begin{matrix} & \begin{matrix} \textcircled{1x} & \textcircled{1y} & \textcircled{3x} & \textcircled{3y} \end{matrix} \\ \begin{matrix} \textcircled{1x} \\ \textcircled{1y} \\ \textcircled{3x} \\ \textcircled{3y} \end{matrix} & \begin{bmatrix} 30.9 & 23.2 & -30.9 & -23.2 \\ 23.2 & 17.4 & -23.2 & -17.4 \\ -30.9 & -23.2 & 30.9 & 23.2 \\ -23.2 & -17.4 & 23.2 & 17.4 \end{bmatrix} \end{matrix}$$



$$\begin{bmatrix} Q_{1x} \\ Q_{1y} \\ Q_{2x} \\ Q_{2y} \\ Q_{3x} \\ Q_{3y} \end{bmatrix} = \begin{bmatrix} \\ \\ \\ \\ \\ \\ \end{bmatrix} \quad \begin{bmatrix} \delta_{1x} \\ \delta_{1y} \\ \delta_{2x} \\ \delta_{2y} \\ \delta_{3x} \\ \delta_{3y} \end{bmatrix}$$

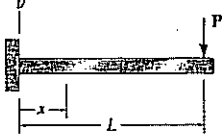
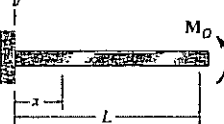
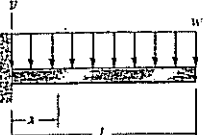
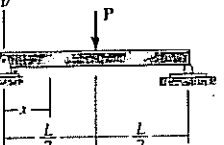
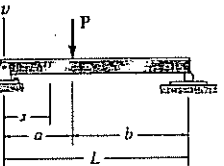
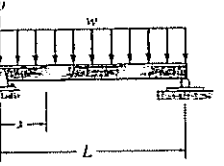
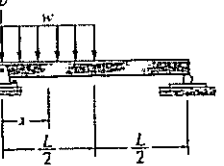
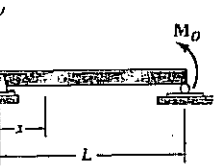
Space for calculations on next page.

**Problem 4 continued.**

Copied from:

Hibbeler, R. C. (1999), Structural Analysis, 4<sup>th</sup> Ed., Prentice Hall, Upper Saddle River, NJ, USA.

### Beam Deflections and Slopes

Loading	$v \uparrow$	$\theta \uparrow$	Equation + $\uparrow$ + $\curvearrowright$
	$v_{max} = -\frac{PL^3}{3EI}$ at $x = L$	$\theta_{max} = -\frac{PL^2}{2EI}$ at $x = L$	$v = -\frac{P}{6EI}(x^3 - 3Lx^2)$
	$v_{max} = \frac{M_0L^2}{2EI}$ at $x = L$	$\theta_{max} = \frac{M_0L}{EI}$ at $x = L$	$v = \frac{M_0}{2EI}x^2$
	$v_{max} = -\frac{wL^4}{8EI}$ at $x = L$	$\theta_{max} = -\frac{wL^3}{6EI}$ at $x = L$	$v = -\frac{w}{24EI}(x^4 - 4Lx^3 + 6L^2x^2)$
	$v_{max} = -\frac{PL^3}{48EI}$ at $x = L/2$	$\theta_{max} = \pm \frac{PL^2}{16EI}$ at $x = 0$ or $x = L$	$v = \frac{P}{48EI}(4x^3 - 3Lx^2), 0 \leq x \leq L/2$
		$\theta_L = -\frac{Pab(L+b)}{6LEI}$ $\theta_R = \frac{Pab(L+a)}{6LEI}$	$v = -\frac{Pbx}{6LEI}(L^3 - b^3 - x^3)$ $0 \leq x \leq a$
	$v_{max} = -\frac{5wL^4}{384EI}$ at $x = \frac{L}{2}$	$\theta_{max} = \pm \frac{wL^3}{24EI}$	$v = -\frac{wx}{24EI}(x^3 - 7Lx^2 + L^3)$
		$\theta_L = -\frac{3wL^3}{128EI}$ $\theta_R = \frac{7wL^3}{384EI}$	$v = -\frac{wx}{384EI}(9L^3 - 24Lx^2 + 16x^3)$ $0 \leq x \leq L/2$ $v = -\frac{wL}{384EI}(8x^3 - 24Lx^2 + 17L^2x - L^3)$ $L/2 \leq x \leq L$
	$v_{max} = -\frac{M_0L^2}{9\sqrt{3}EI}$	$\theta_L = -\frac{M_0L}{6EI}$ $\theta_R = \frac{M_0L}{3EI}$	$v = -\frac{M_0x^2}{6EIL}(x^2 - 3Lx + 2L^2)$